MANAGING

EVEN BRAKE TEMPERATURES

ON TWIN-AISLE AIRPLANES DURING SHORT FLIGHTS BOB MACKNESS ASSOCIATE TECHNICAL FELLOW

BUB MACKNESS ASSOCIATE TECHNICAL FELLOW MECHANICAL/HYDRAULIC SYSTEMS BOEING COMMERCIAL AIRPLANES

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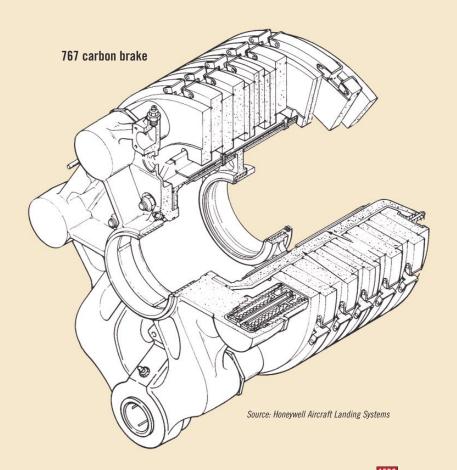


Operators typically purchase twin-aisle airplanes for long-distance flights. However, when market conditions dictate, operators may use some of these airplanes on shorter flights. In such instances, appropriate action by the flight crew can reduce the likelihood of brake overheating and concomitant departure delays.

FLIGHT OPERATIONS



- hen a twin-aisle airplane is used for short-flight operations, brakes may heat unevenly on each landing, with the difference in temperatures becoming more pronounced with repeated landings. At the gate, the brake temperature display may indicate that some brakes are hotter than others. Temperatures may continue to climb until the brake temperature light illuminates or an alert displays. Understanding the causes of uneven brake heating and the meaning of the brake temperature indication can help flight crews minimize possible departure delays. This article explains the following:
- 1. Factors that lead to uneven brake heating.
- 2. Interpretation of the brake temperature indication.
- 3. Flight crew action that can minimize uneven brake heating.



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Slowing an airplane requires the dissipation of kinetic energy. Kinetic energy that is not absorbed by the thrust reversers and speed brakes is converted to heat energy by the wheel brakes. The weight and speed of the airplane during landing and taxiing determine how much energy the brake friction (i.e., heat stack) material absorbs.

Factors that lead to uneven brake heating include variation in brake wear, inadvertent asymmetrical braking, and wind conditions.

Variation in brake wear.

Brakes that are approaching wear limits can become as much as 30 percent hotter than unworn brakes. This variation in temperature simply reflects the difference in the amount of brake friction material.

Inadvertent asymmetrical braking.

Temperature differences between brakes on the left and right gears often arise from inadvertent asymmetrical braking. When the brakes and rudder pedals are used concurrently to maintain heading and keep speed under control, it is easy for flight crews to inadvertently apply asymmetrical braking inputs that unknown to them — are being compensated for by the rudder. When the pedal brakes are applied lightly during taxi braking and landings, temperature differences among the brakes on each landing gear can increase quickly.

Wind conditions.

Brakes cool by convection, and even a moderate breeze can result in temperature differences among brakes that are exposed to wind and those that are sheltered from it.

Given these factors, it is important to remember that temperature differences between the two sides of an airplane do not necessarily mean that a problem exists with the braking system or the brake temperature monitor.

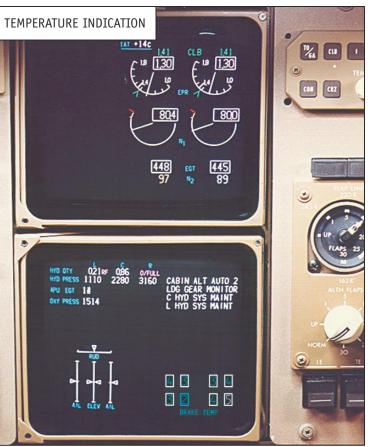
2 INTERPRETATION OF THE BRAKE TEMPERATURE INDICATION

The BRAKE TEMP light illuminates (or a BRAKE OVHT alert displays) when the temperature of a brake rises above a predetermined level and turns off when all the brakes have cooled to a certain level (fig. 1). These levels vary by airplane model and type of brake (table 1).

The BRAKE TEMP light and BRAKE OVHT alert indicate that the radiated and conducted heat from the brake may cause a wheel fuse plug to melt to release air pressure from the tire safely. Whether or not the fuse plug actually melts depends on how far the threshold setting for the brake temperature indication has been exceeded (i.e., the higher the reading on the brake temperature display, the greater the likelihood of the fuse plug melting). The aircraft maintenance manual describes the procedures for checking for deflated tires and hydraulic leaks. (Hydraulic leaks can result from overheated brake

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1	BRAKE TEMPERATURE INDICATION SYSTEMS BY			
Airplane model		Temperature thresholds (°C)		
		Indication on	Indication off	
	7-200/300 :h steel brakes	371	315	■ 1 ■ /
-	7-200/-300 th carbon brakes	427	362	i ? [■ 1
	7-400 th carbon brakes	538	338	■ (1
	7-200/300 th carbon brakes	430	330	■ 1 ■ (
)-11 th carbon brakes	550	450	■ 1 ■ (;



RPLANE MODEL

Notes

The brake temperature monitoring system is optional.

Any brake reading that exceeds the threshold appears in white numerals in a white-bordered box on the engine indication and crew alerting system (EICAS).

The BRAKE TEMP light is white and is not an alert but an aid in the use of the brake cooling chart.

The alert is an amber BRAKE TEMP advisory message on EICAS. On the landing gear synoptic page, any brake reading that exceeds the threshold appears in white numerals in a white-bordered box.

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The BRAKE OVHT alert is a level 2 message on the electronic display. On the configuration page, any brake reading that exceeds the threshold appears in amber numerals inside an amber-bordered box.





Comparison of new (left) and worn carbon brakes

piston seals after a BRAKE TEMP or BRAKE OVHT indication has occurred.) If no anomalies are found, the maintenance crew may clear the airplane for dispatch.

Brake temperatures tend to increase from one landing to the next during shortflight operations. On the first landing of the day, the brakes are relatively cool because of the length of time since the

Source: Messier-Bugatti

last significant energy input. When the airplane leaves the gate for the next flight, however, the brakes will still be warm from the previous landing, and more energy will accumulate from braking during the taxi out. The brakes will not cool completely on a short flight, and during the next landing, the same braking effort will produce higher peak temperatures. The temperature

differences between new and worn brakes also will increase with repeated landings. Because of these factors, a BRAKE TEMP or BRAKE OVHT indication will more likely occur toward the end of the day and especially in hot weather.

If landing conditions for an airplane exceed certain maximums listed in the airplane flight manual (AFM) quickturnaround charts, departure may be delayed for a prescribed period of time to let the brakes cool. The airplane may be allowed to depart before the end of that period if the true brake temperatures, as shown by the brake temperature indication, are lower than those predicted in the AFM.

FLIGHT CREW ACTION THAT CAN MINIMIZE UNEVEN BRAKE HEATING

How the flight crew uses the brake system in conjunction with reverse thrust and the spoilers during landing significantly affects the amount of energy transferred to the brake stacks. Flight crews can use several techniques to arrive at the airport gate with cooler brakes.

On long runways, aerodynamic drag and thrust reversers may provide plenty of deceleration at high speed, and braking can be delayed until taxi speeds are reached. If autobrakes are used, the flight crew should choose the lowest deceleration setting commensurate with the available field length; landing long and turning off short should be avoided. Brake energy balance is improved when the autobrakes are used all the way down to taxi speeds because the same hydraulic pressure is applied to every brake.

On the taxiway, moderate brake applications followed by full pedal release keep the brakes cooler than would a steady application of light brake pressure (i.e., riding the brakes). Using this technique makes it less likely that the BRAKE TEMP or BRAKE OVHT indication will come on because of one disproportionately hot brake. (It should be noted that minimizing the number of brake applications can reduce wear on carbon brakes.)







During short-flight operations, flight crews can take steps to pass on a cooler set of brakes to the next crew. Crews can carefully manage brake energy during landing and taxiing by

- Maximizing the use of thrust reversers and minimizing the use of brakes.
- Avoiding riding the brakes during taxi.

If the BRAKE TEMP or BRAKE OVHT indication still occurs, published maintenance procedures should be performed.

SUMMARY

Using autobrakes or symmetrical pedal braking.

, refer to the Aug. 1, 1990, ew and the latest revision of the Boeing Flight Crew Training Manual.

